TM 55-1510-220-CL

TECHNICAL MANUAL

Operator's and Crewmember's Checklist

ARMY MODEL

RC-12G AIRCRAFT

Pilot's Checklist

This copy is a reprint which includes current pages from Change 1.

HEADQUARTERS,
DEPARTMENT OF THE ARMY

10 December 1985

URGENT

NOTE:

THIS CHANGE HAS BEEN PRINTED AND DISTRIBUTED OUT OF SEQUENCE. IT SHOULD BE INSERTED IN THE MANUAL AND USED. UPON RECEIPT OF THE EARLIER SEQUENCED CHANGE ENSURE A MORE CURRENT CHANGE PAGE IS NOT REPLACED WITH A LESS CURRENT PAGE.

TM 55-1510-220-CL C3

CHANGE NO. 3

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D.C., 7 August 1992

Operator's and Crewmember's Checklist

ARMY MODEL RC-12G AIRCRAFT

Pilot's Checklist

TM 55-1510-220-CL, 10 December 1985, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages

Insert pages

N-9 through N-14

N-9 through N-14

<u>DISTRIBUTION STATEMENT A</u>: Approved for public release; distribution is unlimited.

URGENT

Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

GORDON R. SULLIVAN General, United States Army Chief of Staff

Official: Mitte of danto

MILTON H. HAMILTON Administrative Assistant to the Secretary of the Army 02211

DISTRIBUTION:

To be distributed in accordance with DA Form 12-31-E, block no. 1136, -10 & CL maintenance requirements for TM 55-1510220-CL. CHANGE

HEADQUARTERS
DEPARTMENT OF THE ARMY
WASHINGTON, D.C., 1 November 1990

NO. 1

Operator's and Crewmember's Checklist ARMY MODEL RC-12G AIRCRAFT Pilot's Checklist

TM 55-1510-220-CL, 10 December 1985, is changed as follows:

1. Remove and insert pages as indicated below. New or changed text material is indicated by a vertical bar in the margin. An illustration change is indicated by a miniature pointing hand.

Remove pages Insert pages

E-23/E-24 E-23/E-24

2. Retain this sheet in front of manual for reference purposes.

By Order of the Secretary of the Army:

CARL E. VUONO General, United States Army Chief of Staff

Official:

PATRICIA P. HICKERSON Colonel, United States Army The Adjutant General

DISTRIBUTION:

To be distributed in accordance with DA Form 12-31-E, block no. 1136, -10 & CL maintenance requirements for TM 55-1510-220-CL.

GENERAL INFORMATION AND SCOPE

SCOPE. This checklist contains the operator's and crewmember's checks to be accomplished during normal and emergency operations

GENERAL INFORMATION. The checklist consists of three parts normal procedures, emergency procedures, and performance data. Normal procedures consist of the procedures required for normal flight and those required for "Before Landing". The normal procedures portion will be subdivided to include the before landing checks of Chapter 8 of the Operator's Manual Emergency procedures are subdivided into 7 classifications as follows engine, propeller, (prop), fire, fuel, electrical (elect), landing and ditching (ldg/dtch), and flight controls (fit cont) Performance data consists of performance checks.

NOTE

This checklist does not replace the amplified version of the procedures in the operator's manual (TM 55-1510-220-10), but is a condensed version of each procedure

NORMAL PROCEDURES PAGES. The contents of the normal procedures of this manual are a condensation of the amplified checklist appearing in the normal procedures, or crew duties portion of the applicable operator's manual

EMERGENCY PROCEDURES PAGES. The requirements for this section of the condensed checklist manual (CL) are identical to those for the normal procedures, except that the information is drawn from the amplified checks in the emergency procedures

portion of the operator's manual The emergency requirements are subdivided into the 7 classifications listed above Immediate action items shall be underlined.

Symbols preceding numbered steps

- * Indicates performance of steps is mandatory for all "Thru Flights"
- N Means performance of step is mandatory for "Night Flights".
- ★- Indicates a detailed procedure for this step is included in the Performance Checks section, located at theback of the checklist.
- I- Indicates mandatory check for "Instrument Flights".
- O Indicates if installed
- (3) Copilot duties. To be performed at pilots command. Immediate action emergency items are underlined.

REPORTING ERRORS AND RECOMMENDING IMPROVEMENTS.

You can help improve this manual. If you find any mistakes or if you know of a way to Improve the procedures, please let us know. Mail your Letter, DA Form 2028 (Recommended Changes to Publications and Blank Forms), or DA Form 2028-2 located in the back of the applicable Aircraft Operator's Manual direct to Commander, US Army Aviation Systems Command, ATTN AMSAV-MPSD, 4300 Goodfellow Blvd., St. Louis, MO 63120-1798. A reply will be furnished to you.

NORMAL PROCEDURES

BEFORE EXTERIOR CHECK

- * 1. Publications Check.
- * 2. Oxygen system Check.
- * 3. Flight controls Unlock and check.
- * 4. Parking brake Set.
 - 5. Elevator trim Set to "O" (neutral).
- * 6. Gear- DN.
- * 7. Ice vane pull handles In.
- * 8. Keylock switch ON.
- * 9. Battery switch ON.
- 10. Ice vane switches RETRACT.
- 11. Lighting systems Check as required.
- ★12. Pitot tubes (2), stall warning vane, heated fuel vents (2)Check.
 - 13. Fuel gages Check fuel quantity and gage operation.
 - 14. Battery switch As required.
 - 15. Toilet Check.
 - 16. Emergency equipment Check.
 - 17. Mission equipment and circuit breakers Check and set.
 - 18. Parachutes Check

EXTERIOR CHECK

* FUEL SAMPLE

Check collective fuel sample from all drains for possible contamination.

LEFT WING, AREA 1

1. Left wing area - Check.

LEFT MAIN LANDING GEAR

1. Left main landing gear - Check.

LEFT ENGINE AND PROPELLER

1. Left engine - Check

CENTER SECTION, LEFT SIDE

1. Center section - Check.

FUSELAGE UNDERSIDE

Fuselage underside - Check.

NOSE SECTION, AREA 2

Nose section - Check.

CENTER SECTION, RIGHT SIDE

1. Center section - Check.

RIGHT ENGINE AND PROPELLER

1. Right engine and propeller - Check.

RIGHT MAIN LANDING GEAR

1. Right main landing gear - Check.

RIGHT WING, AREA 3

1. Right wing - Check.

FUSELAGE RIGHT SIDE, AREA 4

1. Fuselage right side - Check.

EMPENNAGE, AREA 5

1. Empennage - Check.

FUSELAGE, LEFT SIDE, AREA 6

1. Fuselage - Check.

N-3

* INTERIOR CHECK

- 1. Cargo/loose equipment Check secure.
- Cabin/cargo doors Test and lock.
- 3. Emergency exit Check secure and key removed.
- 4. Mission cooling ducts Check open and free of obstructions.
- 5. Flare/chaff dispenser preflight test Completed.
- ★6. Crew briefing As required.

BEFORE STARTING ENGINES

- ★* 1. Oxygen system Check as required.
 - 2. Circuit breakers Check In.
 - * 3. Overhead control panel switches Set.
 - * 4. Fuel panel switches Check.
 - 5. Magnetic compass Check.
 - * 6. Pedestal controls Set.
 - * 7. Pedestal extension switches Set.
 - 8. Gear alternate engage and ratchet handles Stowed.
 - 9. Free air temperature gage Check, note current reading.
 - 10. Instrument panel Check and set.
 - PROP SYNC switch OFF.
 - Mission panel switches and circuit breakers Set and OFF.
 - 13. Pressurization controls Set.

- 14. Subpanels Check and set as follows
- Pilot's static air source NORMAL
- 16. Pilot's and copilot's audio control panels As required
- 17. Ice vane pull handles In
- ★18. Fuel pumps/crossfeed operation Check
 - 19. AC and DC GPU As required
 - 20. External power advisory annunciator lights As required
 - DC power Check (22 VDC minimum for battery, 28 maximum for GPU starts)
- ★22. Annunciator panels Test as required
- ★23. Stall and gear warning system Check
- ★24. Fire Protection system Check
 - 25. INS Align as required

* FIRST ENGINE START (BATTERY START)

- 1. Avionics master switch OFF.
- 2. Exterior light switches As required
- 3. Propeller Clear
- 4. Ignition and engine start switch ON
- Condition lever (after N₁ RPM stabilizes, 12% minimum)
 LOW IDLE
 - 6 TGT and N₁ Monitor (TGT 10000C maximum, N₁ 52% minimum)

- 7. Oil pressure Check (60 PSI minimum)
- 8. Ignition and engine start switch OFF, after 50% N
- 9. Condition lever HI IDLE
- 10. Generator switch RESET, then ON

* SECOND ENGINE START (BATTERY START)

- 1. First engine generator load 50% or less
- 2. Propeller Clear
- 3. Ignition and engine start switch ON
- Condition lever (after N1 RPM passes 12% minimum) -LOW IDLE
- 5. TGT and N1 Monitor (TGT 1000°C maximum, N₁ 52% minimum)
- 6. Oil pressure Check (60 PSI minimum)
- 7. Ignition and engine start switch OFF after 50% $\ensuremath{N_{\!\scriptscriptstyle I}}$
- 8. Battery charge light Check
- 9. Inverter switches ON, check INVERTER lights extinguished
- Second engine generator RESET, then ON
- 11. Condition levers As required

ABORT START

- Condition lever FUEL CUTOFF
- 2. Ignition and engine start switch STARTER ONLY
- 3. TGT Monitor for drop in temperature
- 4. Ignition and engine start switch OFF

ENGINE CLEARING

- 1. Condition lever FUEL CUTOFF
- 2. Ignition and engine start switch OFF (5 minute minimum)
- 3. Ignition and engine start switch STARTER ONLY (15 seconds minimum, 30 seconds maximum)
- 4. Ignition and engine start switch OFF

* FIRST ENGINE START (GPU START)

- 1. INS As required
- 2. Avionics master switch As required
- 3. Exterior light switches As required
- 4. Propeller- Clear
- 5. Ignition and engine start switch ON
- Condition lever (after N1, RPM stabilizes, 12% minimum) LOW IDLE
- 7. TGT and N1 Monitor (TGT 10000C maximum, N1 52% minimum)

- 8. Oil pressure Check (60 PSI minimum)
- 9. Ignition and engine start switch OFF after 50% N
- 10. Condition lever- HI IDLE
- 11. DC GPU Disconnect as required
- Generator switch (GPU disconnected) RESET, then ON

* SECOND ENGINE START (GPU START)

- 1. Propeller- Clear
- 2. Ignition and engine start switch ON
- Condition lever (after N1 RPM passes, 12% minimum) -LOW IDLE
- 4. TGT and N1 Monitor (TGT 1000°C maximum, N1 52% minimum).
- 5. Oil pressure Check (60 PSI minimum)
- Ignition and engine start switch OFF, after TGT stabilized
- 7. Propeller levers FEATHER
- 8. GPU Disconnect (Check aircraft external power and mission external power light extinguished)
- 9. Propellers levers HIGH RPM
- Aircraft Inverter switches ON, check #1 INVERTER and #2 INVERTER annunciator lights extinguished
- 11. Generator switches RESET, then ON
- 12. Condition levers As required

BEFORE TAXIING

- * 1. Brake deice As required.
- * 2. Cabin temperature and mode Set.
- * 3. AC/DC power Check.
- * 4. Avionics master switch ON.
 - 5. Mission panel Set and checked as required.
- ★6. Electric elevator trim and autopilot/flight director operation Check.
- ★* 7. Autopilot trim fail system Check.
 - 8. Avionics Check and set as required.
 - 9. Flaps Check.
 - Altimeters Check and set.

*TAXIING

- 1. Brakes- Check.
- 2. Flight instruments Check for normal operation.

ENGINE RUNUP

- 1. Mission control panel Set.
- ★ 2. Propeller manual feathering Check.
- ★ 3. Autofeather Check.
- ★ 4. Overspeed Governors Check.
- ★ 5. Primary governors Check.
- ★ 6. Ice vanes Check.
 - Condition levers HI IDLE.

- 8. Power levers IDLE.
- ★ 9. Anti-ice and deice systems Check.
- ★ 10. Pneumatic pressure Check.
- ★ 11. Pressurization system Check.
 - 12. Condition levers As required.
 - 13. Windshield anti-ice As required.

* BEFORE TAKEOFF

- (1) Autofeather switch ARM.
- (2) Bleed air valves As required.
- (3) Ice and rain switches As required.
- (4) panel Check fuel quantity and switch positions.
- (5) Flight and engine instruments Check for normal indications.
- (6) Cabin altitude and rate-of-climb controller --Set.
- (7) Annunciator panels Check (note indications).
 - 8. Propeller levers HIGH RPM.
- 9. Flaps As required.
- 10. Trim Set.
- 11. Avionics Set.
- Flight controls Check.
- ★ 13. Departure briefing Complete.

* LINE UP

- (1) Transponder As required.
- (2) Engine autoignition switch ARM.
- Power stabilized Check approximately 25% minimum.
- (4) Condition levers LOW IDLE.
- 5. Lights As required.
- Mission control panel Set.

AFTER TAKEOFF

- 1. Gear- UP.
- 2. Flaps UP.
- 3. Landing lights OFF.
- 4. Climb power Set.
- 5. PROP SYNC switch As required.
- (6) Yaw damp As required.
- (7) Autofeather switch As required.
- (8) Brake de-ice As required.
- (9) Windshield anti-ice As required.
- 10. Cabin pressurization Check.
- (11) Wings and nacelles Check.
- (12) Flare/chaff dispenser safety pin (electronic module) Remove.
- (13) Chaff function selector switch As required.

N-11 C3

CRUISE

- 1. Power Set.
- 2. Ice and rain switches As required.
- (3) Auxiliary fuel gages Monitor.
- (4) Altimeters Check.
- (5) Engine instrument indications Check.
 - 6. Recognition lights As required.

DESCENT - MAX RATE (CLEAN)

- (1) Cabin pressurization Set.
- 2. Power levers IDLE.
- 3. Propeller levers HIGH RPM.
- 4. Flaps UP.
- 5. Gear- UP.
- 6. Airspeed Vmo.
- (7) Ice and rain switches As required.
- 8. Recognition lights As required.

DESCENT - MAX RATE (LANDING CONFIGURATION)

- (1) Cabin pressurization Set.
- 2. Power levers- IDLE.
- 3. Propeller levers HIGH RPM.
- 4. Flaps APPROACH.
- 5. Gear- DN.
- 6. Airspeed 184 KIAS maximum.
- (7) Ice & rain switches As required.
- 8. Recognition lights As required.

DESCENT-ARRIVAL

- (1) Cabin pressurization Set.
- (2) Ice and rain switches As required.
- (3) Windshield anti-ice As required.
- 4. Recognition lights ON.
- 5. Altimeters Set to current altimeter setting.
- (6) Flare/chaff dispenser arm-safe switch -SAFE.
- (7) Flare/chaff dispenser safety pin (electronic module) -Insert.
- ★ 8. Crew briefing Complete.

BEFORE LANDING

- 1. Propeller synchronization switch -OFF.
- (2). Autofeather switch ARM.
 - 3. Propeller levers As required.
 - 4. Flap switch (below 202 KIAS) APPROACH.
 - 5. Gear DN.
 - 6. Landing lights As required.
- (7). Brake deice As required.

LANDING

- 1. Autopilot and yaw damp Disengaged.
- 2. Gear down lights Check three green.
- 3. Propeller levers HIGH RPM.

TOUCH-AND-GO LANDINGS

- (1). Propeller levers HIGH RPM.
- (2). Flaps As required.
- (3). Trim Set.
 - 4. Power stabilized Check approximately 25% minimum
 - 5. Takeoff power Set

GO-AROUND

- 1. Power As required.
- 2. Gear UP.
- 3. Flaps UP.
- 4. Landing lights OFF.
- 5. Climb power Set.
- (6). Yaw damp As required.
- (7). Brake deice OFF

AFTER LANDING

- (1). Condition levers As required.
- (2). Engine autoignition switch OFF.
- (3). Ice and rain switches OFF.
- (4). Flaps UP.
- (5). Transponder As required.
 - 6. Lights As required.
- (7). Mission control panel Set.

N-16

ENGINE SHUTDOWN

- 1. Brake deice OFF.
- 2. Parking brake Set.
- 3. Landing/taxi lights OFF.
- 4. Cabin temperature mode selector switch OFF.
- 5. Autofeather switch OFF.
- 6. Vent and aft vent blower switches AUTO.
- 7. INS OFF.
- Mission equipment OFF, as required.
- 9. Inverter switches OFF.
- 10. Battery condition Check as required.
- 11. TGT Check.
- 12. Propeller levers FEATHER.
- 13. Condition levers FUEL CUTOFF.
- 14. Exterior lights OFF.
- 15. Master panel lights switch OFF.
- 16. Avionics master switch Off.
- 17. Master switch OFF.
- 18. Keylock switch OFF.
- 19. Oxygen system OFF.

BEFORE LEAVING AIRCRAFT

- 1. Wheels Chocked.
- 2. Parking brake As required.
- 3. Flight controls Locked.
- 4. Overhead flood lights Off.
- 5. Standby fuel pump switches OFF.
- 6. Transponder OFF.
- 7. Mode 4 As required.
- 8. Emergency exit lock As required.
- 9. Aft cabin light OFF.
- 10. Door light OFF.
- 11. Walk-around Inspection Complete.
- 12. Aircraft forms Complete.
- 13. Aircraft secured.

N-18

EMERGENCY PROCEDURES

NOTE

The urgency of certain emergencies requires Immediate and Instinctive action by the pilot. The most Important single consideration is aircraft control. All procedures are subordinate to this requirement.

ENGINE MALFUNCTION

Engine Malfunction Before Liftoff (Abort)

- 1. Power levers IDLE.
- 2. Braking As required.
- (3). Condition levers FUEL CUTOFF.
- (4). Fire pull handles Pull.
- (5). Master switch OFF.

E-1

Engine Malfunction After Liftoff (Abort)

- 1. Power levers Reduce.
- 2. Gear- DN.
- 3. Complete a normal landing.

NOTE

If able to land on remaining runway, check gear down and use brakes and reverse thrust as required. If Insufficient runway remains for stopping, perform the following.

- (4). Condition levers FUEL CUTOFF.
- (5). Fire pull handles Pull
- (6). Master switch OFF

Engine Malfunction After Liftoff (Flight Continued)

- 1. Power Maximum allowable.
- 2. Gear UP.
- 3. Flaps UP.
- 4. Landing light OFF.
- 5. Brake deice OFF.
- 6. Engine cleanup Perform.

Engine Malfunction During Flight

- 1. Autopilot/yaw damp DISENGAGE.
- 2. Power As required.
- 3. Dead engine Identify.
- 4. Power lever (dead engine) IDLE.
- 5. Propeller lever (dead engine) FEATHER.
- 6. Propeller synchronization switch OFF.
- 7. Gear As required.
- 8. Flaps As required.
- 9. Power Set for single engine cruise.
- (10). Engine cleanup Perform.

Engine Malfunction During Final Approach

- 1. Power As required.
- 2. Gear DN.

Engine Malfunction (Second Engine)

- 1 Airspeed 140 KIAS.
- 2. Power lever IDLE.
- Propeller lever Do not FEATHER.
- 4. Conduct engine restart procedure.

Engine Shutdown In Flight

- 1. Power lever IDLE.
- 2. Propeller lever- FEATHER.
- 3. Condition lever FUEL CUTOFF.
- (4). Engine cleanup Perform.

Engine Cleanup

- (1). Autoignition switch OFF.
- (2). Autofeather switch OFF.
- (3). Generator switch OFF.
 - 4. Propeller synchronization switch OFF.

E-4

Engine Restart During Flight Using Starter

- (1). Cabin temperature mode selector switch OFF.
- (2). Electrical load Reduce to minimum.
 - 3. Fire pull handle In.
 - 4. Power lever IDLE.
 - 5. Propeller lever FEATHER.
 - 6. Condition lever FUEL CUTOFF.
 - 7. TGT (operative engine) 700°C or less.
- (8). Ignition and engine start switch ON.
 - 9. Condition lever LOW IDLE.
- 10. TGT Monitor (1,0000C for 5 seconds maximum).
- 11. Oil pressure Check.
- (12). Ignition and engine start switch OFF at 50% $N_{\rm l}$.
- (13). Generator switch RESET, then ON.
- (14). Engine cleanup Perform If engine restart unsuccessful.
- (15). Cabin temperature mode selector switch As required.
- (16). Electrical equipment As required.
- (17). Autoignition switch ARM.
 - 18. Propellers Synchronize.
 - 19. Power As required.

Engine Restart During Flight (Not Using Starter)

- (1). Cabin temperature mode selector switch OFF.
- (2). Electrical load Reduce to minimum.
- (3). Generator switch (affected engine) OFF.
 - 4. Fire pull handle Check in.
 - Power lever IDLE.
 - 6. Propeller lever HIGH RPM.
 - 7. Condition lever FUEL CUTOFF.
 - 8. Airspeed 140 KIAS minimum.
 - 9. Altitude below 20.000 feet Check.
- (10). Engine autoignition switch ARM.
 - 11. Condition lever LOW IDLE.
 - 12. TGT Monitor (1,000°C for 5 seconds maximum).
 - 13. Oil pressure Check.
- (14). Generator switch RESET then ON.
- (15). Engine Cleanup Perform If engine restart unsuccessful.
- (16). Cabin temperature mode selector switch As required.
- (17). Electrical equipment As required.
- (18). Autoignition switch ARM.
 - 19. Propellers Synchronized.
 - 20. Power As required

Low Oil Pressure

- 1. Oil pressure below 105 PSI below 21,000 feet or 85 PSI 21,000 feet and above, torque 49% maximum.
- 2. 011 pressure below 60 PSI Perform engine shutdown, or land as soon as practicable using minimum power to insure safe arrival.

Chip Detector Warning Light Illuminated

If a L CHIP DETR or a R CHIP DETR warning light illuminates, and safe single-engine flight can be maintained, perform engine shutdown

DUCT OVERTEMP CAUTION ANNUNCIATOR LIGHT ILLUMINATED

- (1). Cabin air control In.
- (2). Cabin temperature mode selector switch AUTO.
- (3). Cabin temperature control rheostat Full decrease.
- (4). Vent blower switch HI.
- (5). Cabin temperature mode selector switch MAN COOL.
- (6). Manual temperature switch DECREASE (hold).
- (7). Left bleed air valve switch ENVIRO OFF.
- (8). If the light Is still Illuminated after 30 seconds: Left bleed air valve switch OPEN.
- (9). Right bleed air valve switch ENVIRO OFF.

(10). If the light Is still Illuminated after 30 seconds: Right bleed air valve switch - OPEN.

ICE VANE FAILURE

- 1. Airspeed 160 KIAS or below.
- (2). Ice vane control circuit breaker Pull.
 - 3. Ice vane Operate manually.
 - 4. Airspeed Resume normal airspeed.

BLEED AIR SYSTEM FAILURE

Bleed Air Failure Light Illuminated

- (1). Brake deice switch OFF.
- (2). TGT and torque Monitor (note readings).
- (3). Bleed air valve switch PNEU & ENVIRO OFF.
- (4). Cabin pressurization Check.

Excessive Differential Pressure

- (1). Cabin altitude and rate-of-climb controller. Select higher setting.
- (2). If condition persists LEFT BLEED AIR VALVE switch ENVIRO OFF (light illuminated).
- If condition still persists RIGHT BLEED AIR VALVE switch - ENVIRO OFF (light Illuminated).
 - 4. If condition still persists Descend Immediately.
- (5). If unable to descend CABIN PRESS DUMP switch -CABIN PRESS DUMP.
- (6). Bleed air valve switches OPEN, If cabin heating is required.

LOSS OF PRESSURIZATION (ABOVE 10,000 FEET)

- 1. Crew oxygen masks 100% and on.
- (2). Passenger masks On and check.

CABIN DOOR CAUTION LIGHT ILLUMINATED

- (1). Bleed air valve switches ENVIRO OFF.
 - 2. Descend below 14,000 feet as soon as practicable.
 - 3. Oxygen As required.

SINGLE-ENGINE DESCENT/ARRIVAL

- (1). Cabin controller Set.
- (2). Ice and rain switches As required.
 - 3. Altimeters Set.
 - 4. Recognition lights ON.
- ★5. Arrival briefing Complete.

SINGLE-ENGINE BEFORE LANDING

- 1. Propeller lever As required.
- 2. Flaps- APPROACH.
- 3. Gear- DN.
- 4. Landing lights As required.
- (5). Yaw damp OFF.
- (6). Brake deice OFF.

SINGLE-ENGINE LANDING CHECK

- 1. Autopilot/yaw damp Disengaged.
- 2. Gear lights Check (three green).
- 3. Propeller lever (operative engine) HIGH RPM.

SINGLE-ENGINE GO-AROUND

- 1. Power Maximum allowable.
- 2. Gear- UP.
- 3. Flaps As required.
- 4. Landing lights OFF.
- 5. Power As required.
- (6). Yaw damp As required.

PROPELLER FAILURE (OVER 2080 RPM)

- 1. Power lever (affected engine) IDLE.
- 2. Propeller lever FEATHER.
- 3. Condition lever As required.
- (4). Engine cleanup As required.

E-11/(E-12 Blank)

FIRE

Engine/Nacelle Fire During Start or Ground Operations

- 1. Propeller levers FEATHER.
- 2. Condition levers FUEL CUTOFF.
- 3. Fire pull handle Pull.
- 4. Push to extinguish switch Push.
- 5. Master switch OFF.

Engine Fire In Flight (Fire Pull Handle Light Illuminated)

- 1. Power lever IDLE.
- 2. If fire pull handle light out is extinguished Advance power.
- 3. If fire pull handle light Is still Illuminated Engine fire in flight procedures (identified) Perform.

Engine Fire In Flight (Identified)

- 1. Power lever IDLE.
- 2. Propeller lever FEATHER.
- 3. Condition lever FUEL CUTOFF.
- 4. Fire pull handle Pull.
- 5. Fire extinguisher Actuate as required.
- (6). Engine cleanup Perform.

Fuselage Fire

- 1. Fight the fire.
- 2. Land as soon as possible.

Wing Fire

- 1. Perform engine shutdown on affected side.
- 2. Land as soon as possible.

Electrical Fire

- 1. Crew oxygen 100%.
- Master switch OFF (visual conditions only).
 - 3. All nonessential electrical equipment OFF.
- (4). Battery switch ON.
- (5). Generator switches (Individually) RESET, then ON.
- (6). Circuit breakers Check for Indication of defective circuit.
- (7). Essential electrical equipment On (individually until fire source is Isolated).
 - 8. Land as soon as practicable.

Smoke and Fume Elimination

- 1. Crew oxygen 100% and ON.
- (2). Bleed air valve switches ENVIRO OFF.
- (3). Vent blower switch AUTO.
- (4). Aft vent blower switch OFF.
- Cabin temperature mode selector switch OFF.
- (6). If smoke and fumes are not eliminated. Cabin pressure dump switch CABIN PRESS DUMP.
 - 7. Engine oil pressure Monitor.

FUEL SYSTEM

Fuel Pressure Warning Annunciator Light Illuminated

- (1). Standby pump switch ON.
- (2). Fuel pressure warning annunciator light Check extinguished.
- (3). If fuel pressure warning light Is still illuminated. Record unboosted time.

No Fuel Transfer Caution Annunciator Light Illuminated

- (1). AUX TRANSFER switch (affected side) OVERRIDE.
- (2). Auxiliary fuel quantity Monitor.
- (3). AUX TRANSFER switch (after respective auxiliary fuel has completely transferred) AUTO.

Nacelle Fuel Leak

- 1. Perform engine shutdown.
- 2. Fire pull handle Pull.
- 3. Land as soon as practicable.

Fuel Crossfeed

- (1). AUX TRANSFER switches AUTO.
- (2). Standby pumps OFF.
- (3). Crossfeed switch As required.
- (4). Fuel crossfeed advisory annunciator light Check Illuminated
- (5). Fuel pressure light extinguished Check.
- (6). Fuel quantity Monitor.

Illumination of The #1 NAC LOW or #2 NAC LOW Caution Annunciatorlight.

- 1. Twenty minutes fuel remaining Confirm.
- 2. Land as soon as possible.

E-16

ELECTRICAL SYSTEM

DC Generator Caution Annunciator Light Illuminated

- (1). Generator switch OFF, RESET, then ON.
- (2). Generator switch (no reset) OFF.
- (3). Mission control switch OVERRIDE.
- (4). Operating loadmeter 100% maximum.

Both DC Generator Warning Annunciator Lights Illuminated

- (1). All nonessential equipment OFF.
- (2). Land as soon as practicable.

Excessive Loadmeter Indication (Over 100%)

- (1). Battery switch OFF (monitor loadmeter).
- (2). Loadmeter over 100% Nonessential electrical equipment OFF.
- (3). Loadmeter under 100% BATT switch ON.

Inverter Caution Annunciator Light Illuminated

(1). Affected #1 INVERTER or #2 INVERTER switch - OFF.

INST AC Warning Annunciator Light Illuminated

- N₁ and TGT Indications Check.
- 2. Other engine instruments Monitor.

Circuit Breaker Tripped

- (1). BUS FEEDER breaker tripped Do not reset.
- (2). Nonessential circuit Do not reset.
- (3). Essential circuit Reset once.

Battery Charge Caution Annunciator Light Illuminated

- (1). Loadmeter Check; note indication.
- (2). Battery switch OFF
- (3). Loadmeter Check If loadmeter indicates less that 2 5% change (one needle width), turn battery switch ON and monitor for increasing load If load continues to increase, turn battery switch OFF
- (4) Battery switch (landing gear/flap extension only) ON

EMERGENCY DESCENT

- 1. Power lever IDLE.
- 2. Propeller lever HIGH RPM.
- 3. Flaps APPROACH.
- 4. Gear DN.
- 5. Airspeed 184 KIAS maximum.

LANDING EMERGENCIES

Landing Gear Unsafe Indication

- 1. Gear DN.
- 2. Gear lights Check (three green).
- (3). Landing gear relay circuit breaker Check in.

Landing Gear Emergency Extension

- 1. Airspeed 130 KIAS.
- (2). LANDING GEAR RELAY circuit breaker Out.
 - 3. Gear handle DN.
 - 4. Landing gear alternate engage handle Lift and turn clockwise to the stop.
 - 5. Alternate landing gear extension handle Pump.
 - 6. Gear lights Check (three green).

Gear-Up Landing (All Gear Up or Unlocked)

- (1). Crew emergency briefing Complete.
- (2). Loose equipment Stowed.
- (3). Bleed air valves ENVIRO OFF.
- (4). Cabin pressure dump switch CABIN PRESS DUMP.
- (5). Cabin emergency hatch Remove and stow.
 - 6. Seat belts and harnesses Secured.
 - 7. Landing gear alternate engage handle Disengaged.
 - 8. Alternate landing gear extension handle Stowed.
- (9). Gear relay circuit breaker In.
- 10. Gear UP.
- 11. Nonessential electrical equipment OFF.
- 12. Flaps As required (DOWN for landing).
- 13. Power levers (runway assured) IDLE.
- (14). Condition levers FUEL CUTOFF.
- (15). Fire pull handles Pull.
- (16.). Master switch OFF.

Landing With Nose Gear Unsafe

- (1). Crew emergency briefing Complete.
- (2). Loose equipment Stowed.
- (3). Bleed air valves ENVIRO OFF.
- (4). Cabin pressure dump switch CABIN PRESS DUMP.
- (5). Cabin emergency hatch Remove and stow.
 - 6. Seat belts and harnesses Secured.
- (7). Nonessential electrical equipment OFF.
 - 8. Power levers (runway assured) IDLE.
- (9). Condition levers FUEL CUTOFF.
- (10). Fire pull handle Pull.
- (11). Master switch OFF.

Landing With One Main Gear Unsafe

- (1). Crew emergency briefing Complete.
- (2). Loose equipment Stowed.
- (3). Bleed air valve switches ENVIRO OFF.
- (4). Cabin pressure dump switch CABIN PRESS DUMP.
- (5). Cabin emergency hatch Remove and stow.
 - 6. Seat belts and harnesses Secured.
- (7). Nonessential electrical equipment OFF.
 - 8. Touchdown On safe main gear first.
 - 9. Power levers (runway assured) IDLE.
- (10). Condition levers FUEL CUTOFF.
- (11). Fire pull handle Pull.
- (12). Master switch OFF.

CRACKED WINDSHIELD

External Crack

No action is required in flight.

Internal Crack

- Descend to below 25,000 feet
- Cabin Pressure Reset pressure differential to 4 PSI or less within 10 minutes

CRACKED CABIN WINDOW

- 1. Oxygen As required.
- 2. Cabin pressurization Depressurize.
- 3. Descend As required.

DITCHING

- (1). Radio calls/transponder As required.
- (2). Crew emergency briefing As required.
- (3). Bleed air valves ENVIRO OFF.
- (4). Cabin pressure dump switch CABIN PRESS DUMP.
- (5). Cabin emergency hatch Remove and stow.
 - 6. Seat belts and harnesses Secured.
 - 7. Gear UP.
 - 8. Flaps DOWN.
- (9). Nonessential electrical equipment OFF.
- 10. Approach Normal, power on.
- (11). Emergency lights As required.

E-23(E-24 Blank) C1

FLIGHT CONTROLS MALFUNCTION

Autopilot/Yaw Damper Emergency Disconnection:

The autopilot can be disengaged by any of the following methods.

- 1. Pressing the DISC TRIM AP YD disconnect switch (control wheels).
- 2. Moving the autopilot engage-disengage switch (autopilot mode selector control panel to the DIS position.
- 3. Pressing the go-around switch (left power lever), (yaw damper will remain on).
- 4. Pulling the AP PWR and AFCS DIRECT circuit breakers (overhead control panel).
- 5. Setting AVIONICS MASTER PWR switch (overhead control panel) to the OFF position.
- 6. Setting aircraft MASTER switch (overhead control panel) to the OFF position.

Unscheduled Rudder Boost Activation

(1). Rudder boost - OFF.

If condition persists

- (2). Bleed air valve PNEU & ENVIRO OFF.
 - 3. Rudder trim Adjust.

Unscheduled Electric Elevator Trim

- Elevator trim switch OFF.
- (2). Elevator trim circuit breaker Out.

E-25/(E-26 Blank)

BAILOUT

- 1. Notify crew to prepare to ball out.
- (2). Distress message Transmit.
- (3). Voice security ZEROIZE.
- (4). Transponder 7700.
 - 5. Flaps DOWN.
 - 6. Airspeed 100 KIAS.
 - 7. Trim As required.
 - 8. Autopilot Engage.
- (9). Cabin pressure switch DUMP.
- 10. Parachute Attach to harness.
- 11. Cabin door Open.
- 12. Abandon the aircraft.

E-27/(E-28 Blank)

PERFORMANCE CHECKS

PITOT TUBES (2), STALL WARNING VANE, HEATED FUEL VENTS (2)

- 1. Stall warning heat switch ON.
- 2. Pitot heat switches (2) ON Check cover removed.
- 3. Fuel vent heat switches (2) ON.
- Left wing heated fuel vent Check by feel for heat and condition.
- 5. Stall warning vane Check by feel for heat and condition.
- Left pitot tube Check by feel for heat and free of obstructions.
- 7. Right pitot tube Check by feel for heat and free of obstructions.
- 8. Right wing heated fuel vent Check by feel for heat and condition.
- 9. Stall warning heat switch OFF.
- 10. Pitot heat switches (2) OFF.
- 11. Heated fuel vent switches (2) OFF.

FIRE EXTINGUISHER PRESSURE

Temp°C	-40	-29	-18	-06	04	16	27	38	۷
·	190	220	250	290	340	390	455	525	6
PSI	to	t							
	240	275	315	365	420	480	550	635	7

OXYGEN SYSTEM

- 1. Oxygen supply pressure gages Check.
- 2. Supply control lever (green) ON.
- 3. Diluter control lever 100% OXYGEN.
- 4. Emergency control lever (red) Set to TEST MASK position while holding mask directly away from face, then return to NORMAL.
- 5. Oxygen masks Put on and adjust.
- Emergency pressure control lever Set to TEST MASK position and check mask for leaks, then return lever to NORMAL.
- Flow indicator Check During inhalation blinker appears, during exhalation blinker disappears. Repeat a minimum of 3 times.

FUEL PUMPS/CROSSFEED

- Fire pull handles Pull.
- 2. Standby fuel pump switches ON.
- 3. Battery switch ON.
- #1 fuel pressure and #2 fuel pressure warning lights -Illuminated.
- 5. Fire pull handles In.
- 6. #1 fuel press and #2 fuel press warning annunciator lights Extinguished.
- 7. Standby fuel pump switches OFF.
- #1 fuel pressure and #2 fuel pressure warning lights -Illuminated.
- Crossfeed Check. Check system operation by activating switch momentarily left then right, noting that #1 FUEL PRESS and #2 FUEL PRESS warning annunciator lights extinguish and that the FUEL CROSS-FEED advisory annunciator light illuminates as switch is energized.

ANNUNCIATOR PANELS

 MASTER CAUTION, MASTER WARNING, #1 FUEL PRESS, #2 FUEL PRESS, GEAR DN, L BL AIR FAIL, R BL AIR FAIL, INST AC, #1 DC GEN, #1 INVERTER, #1 NO FUEL XFR, #2 NO FUEL XFR, #2 INVERTER, #2 DC GEN, Check illuminated.

- ANNUNCIATOR TEST switch Press and hold Check that the annunciator panels, FIRE PULL handle lights, marker beacon lights, MASTER CAUTION and MASTER WARNING lights are Illuminated. Release switch and check that all lights except those in step 1 are extinguished.
- 3. MASTER CAUTION and MASTER WARNING lights Press Check that both lights extinguish.

STALL AND GEAR WARNING SYSTEM

- 1. STALL WARN TEST switch TEST Check that warning horn sounds.
- 2. LDG GEAR WARN TEST switch TEST Check that warning horn sounds and that the LDG GEAR CONTR handle warning lights (2) Illuminate.

FIRE PROTECTION SYSTEM

- Fire Detector Test switch Rotate counterclockwise to check three DETR positions FIRE PULL handles should illuminate in each position. Reset MASTER WARNING in each position.
- Fire Detector Test switch Rotate counterclockwise to check two EXTGH positions. SQUIB OK light, associated #1 EXTGH DISCH and #2 EXTGH DISCH annunciator caution light and MASTER CAUTION LIGHT should illuminate in each position.

ELECTRIC ELEVATOR TRIM AND AUTOPILOT/FLIGHT DIRECTOR OPERATION

- Pilot's and copilot's PITCH TRIM switches Press to NOSE UP and NOSE DN positions, singularly and in pairs. Check that trim wheel moves in proper direction and operates only when trim switches are pressed in pairs. Any deviation requires that electric elevator trim be turned off and flight conducted using manual trim.
- DISC TRIM switch Press to second detent and verify that electric trim disconnects and ELEV TRIM annunciator light (in pedestal) extinguishes.
- 3. Flight director (FDI) and Radio Magnetic Indicator (RMI) warning flags Check masked.

NOTE

Since the pressure of airflow that normally opposes movement of control surfaces is absent during preflight check, it is possible to get a hard over control surface deflection if an autopilot command is allowed to remain active for any appreciable length of time. Move turn knob and pitch thumbwheel only enough to check operation, then return them to the center position.

- 4. Autopilot mode selector panel Select HDG mode.
- Horizontal Situation Indicator (HSI) Set heading marker under lubber line.
- Autopilot Engage Check that controls are stiff and that AP ENG, HDG, and AIL HI TORQUE annunciator lights illuminate.

- 7. AIL HI TORQUE test switch Engage Check that AIL HI TORQUE light extinguishes.
- 8. HSI heading marker Move 10° left and right and verify that flight director and control wheels respond in the appropriate direction.
- Autopilot/yaw damp disengage switch (control wheels) -Press to first detent and verify that autopilot disengages (AP DISC annunciator illuminates) and that flight controls are free.
- 10. Autopilot Engage.
- Autopilot pitch-turn control (pedestal extension) -Command 5° trim UP and verify that manual trim wheel moves nose up and AP trim light indicates UP trim.
- Pitch trim switch (control wheels) Command nose down and verify that autopilot disengages and AP TRIM FAIL and MASTER WARNING annunciator lights illuminate

NOTE

The AP TRIM FAIL annunciator will extinguish by pressing the AP/YD disconnect button on the control wheel to the second detent.

- Repeat steps 9 thru 12 above using opposite commands.
- 14. Autopilot- Engage.
- 15. HSI heading marker Move to command a bank on flight director Indicator.
- GO-AROUND switch Press and verify that GA annunciator light illuminates, autopilot disengages, and that flight director Indicator commands a wings-level 7° nose-up attitude.

17. TEST switch (pilot's flight director indicator) - Press and verify that attitude display indicates an additional 10° pitch up and 20° right roll, and GYRO flag is visible

AUTOPILOT TRIM FAIL SYSTEM

- Autopilot Engage Command DN on AP pitch wheel and hold TRIM TEST switch when elevator trim wheel starts to rotate.
- 2. Verify that autopilot disengages and AP TRIM FAIL and MASTER WARNING lights illuminate within 10 seconds. Repeat steps 1 and 2 using opposite commands.

PROPELLER MANUAL FEATHERING

- 1. Condition lever LOW IDLE.
- Left propeller lever FEATHER Check that propeller feathers.
- 3. Left propeller lever HIGH RPM.
- 4. Repeat procedure for right propeller.

AUTOFEATHER

- 1. Condition levers LOW IDLE.
- 2. Autofeather switch Hold to TEST. (#1 AUTO-FEATHER and #2 AUTOFEATHER advisory annunciator lights should remain extinguished).
- Power levers Advance to approximately 22% torque, then move autofeather switch to TEST Both #1 AUTOFEATHER and #2 AUTOFEATHER advisory annunciator lights should illuminate.

- 4. Left power lever Retard.
 - a. At approximately 16 to 21% torque, check #2 AUTOFEATHER advisory annunciator extinguished.
 - b. At approximately 9 to 14% torque, check #1 AUTOFEATHER advisory annunciator light extinguished (Left propeller starts to feather).
- 5. Left power lever Set approximately 22% torque.
- 6. Repeat steps 1 through 4 for right engine.

OVERSPEED GOVERNORS

- 1. Power levers Set approximately 1950 RPM (both engines).
- #1 propeller governor test switch Hold to TEST position.
- 3. #1 propeller RPM 1830 to 1910 Check.
- 4. Repeat steps 2 and 3 for # 2 engine.
- Power levers Set 1800 RPM.

PRIMARY GOVERNORS

- 1. Power levers Set 1800 RPM.
- 2. Propeller levers Move aft to detent. Check that propeller RPM drops to 1600 to 1640 RPM
- 3. Propeller levers HIGH RPM.

ICE VANES

- Ice vane switches EXTEND Verify torque drop, TGT increase, and #1 ICE VANE EXTEND and #2 ICE VANE EXTEND annunciators illuminate.
- 2. Ice vane switches RETRACT Verify return to original torque and TGT, and that #1 ICE VANE EXTEND and #2 ICE VANE EXTEND annunciators extinguish.

ANTI-ICE AND DEICE SYSTEMS

- Windshield anti-ice switches NORMAL and HI Check PILOT and COPILOT (individually) for loadmeter rise, then OFF.
- 2. Propeller switches INNER and OUTER (momentarily) Check for loadmeter rise.
- Surface deice switch SINGLE CYCLE AUTO. Check for a drop in pneumatic pressure and wing deice boot inflation and after 6 seconds for a second drop in pneumatic pressure.
- 4. Surface deice switch MANUAL Check that surface boots inflate, and remain inflated, then OFF.
- 5. Antenna deice switch SINGLE Check for a drop in pneumatic pressure and antenna deice boot inflation.
- 6. Antenna deice switch MANUAL Check that boots inflate, and remain inflated, then OFF

- Engine Inlet lip heat switches ON. Check that #1 LIP HEAT ON and and #2 LIP HEAT ON advisory annunciator lights are Illuminated, and the #1 LIP HEAT and #2 LIP HEAT caution annunciator lights are extinguished, then OFF.
- 8. Anti-ice and deice system switches OFF

PNEUMATIC PRESSURE

- 1. Power levers IDLE.
- 2. Left bleed air valve switch PNEU & ENVIRO OFF.
- 3. Pneumatic pressure Check 12 to 20 PSI.
- Right pneumatic and environmental switch PNEU & ENVIRO OFF Check that L BL AIR FAIL and R BL AIR FAIL annunciator lights , and L BLAIR OFF and R BLAIR OFF annunciator lights are illuminated.
- 5. Pneumatic pressure Verify zero.
- Left pneumatic and environmental switches OPEN. Check that L BL AIR FAIL and R BL AIR FAIL annunciator lights, and L BL AIR OFF and R BL AIR OFF annunciator lights are extinguished.
- 7. Pneumatic pressure Verify 12 to 20 PSI.
- 8. Right pneumatic and environmental switches OPEN.

PRESSURIZATION SYSTEM

- 1. Cabin door caution light Check extinguished.
- Storm windows Check closed.
- 3. Bleed air valve switches Check OPEN.

- 4. Cabin altitude Set 500 feet lower than airfield elevation.
- 5. Cabin pressure/dump switch TEST (hold).
- 6. Cabin rate-of-climb gage Check for descending Indication and, when confirmed, release cabin pressure/dump switch from TEST.
- Aircraft altitude Set to planned cruise altitude plus 500 feet. (If this setting does not result In a CABIN ALT Indication of at least 500 feet over takeoff field pressure altitude, adjust as required).
- 8. Rate control Set between 9 and 12 o'clock.

CREW BRIEFING.

Crew introduction.

Equipment.

- 1. Personal to include ID tags.
- 2. Professional (medical equipment, etc.).
- Survival.

Flight data.

- 1. Route.
- 2. Altitude.
- 3. Time enroute.
- 4. Weather.

Normal procedures.

- 1. Entry and exit of aircraft.
- 2. Seating and seat position.

- 3. Seat belts.
- 4. Movement in aircraft.
- 5. Internal communications.
- 6. Security of equipment.
- 7. Smoking.
- 8. Oxygen.
- 9. Refueling.
- 10. Weapons and prohibited items.
- 11. Protective masks.
- 12. Toilet.

Emergency procedures.

- 1. Emergency exits.
- 2. Emergency equipment.
- 3. Emergency landing/ditching procedures.

DEPARTURE BRIEFING.

ATC clearance - Review.

- 1. Routing.
- 2. Initial altitude.

Departure procedure - Review.

- 1. SID.
- 2. Noise abatement procedure.
- 3. VFR departure route.

Approach procedure - Review.

- 1. Approach plan/profile.
- 2. Altitude restrictions.
- 3. Missed approach.
 - a. Point.
 - b. Time.
 - c. Intentions.
- 4. Decision height or MDA.
- 5. Lost communications.

Back up approach/frequencies.

Copilot duties - Review.

- 1. Nav/Com set-up.
- 2. Monitor altitude and airspeeds.
- 3. Monitor approach.
- 4. Call out visual/field in sight.

Landing performance data - Review.

- 1 Approach speed.
- 2 Runway required.

Copilot duties - Review.

- 1. Adjust takeoff power.
- 2. Monitor engine instruments.
- 3. Power check at 65 knots.
- 4. Call out engine malfunctions.
- Tune/ident all nav/com radios.
- 6. Make all radio calls.
- 7. Adjust transponder and radar as required.
- 8. Complete flight log during flight (note altitudes and headings).
- 9. Note departure time.

PPC - Review.

- 1. Takeoff power.
- 2. V_r.
- 3. V_y (climb to 500' AGL).
- 4. V_{yse}.

ARRIVAL BRIEFING.

Weather/altimeter setting.

Airfield/facilities - Review.

- 1. Field elevation.
- 2. Runway length.
- 3. Runway condition.

By Order of the Secretary of the Army:

JOHN A. WICKHAM, JR. General, United States Army Chief of Staff

Official:

MILDRED E. HEDBERG Brigadier General, United States Army The Adjutant General

DISTRIBUTION:

To be distributed in accordance with DA Form 12-31, -10 & CL Maintenance requirements for RC-12D Airplane, Reconnaissance.

*GPO: 1994 0 - 382-492

RECOMMENDED CHANGES TO EQUIPMENT TECHNICAL PUBLICATIONS

	' '												
7		,)			SOMET		WRONG WITH PUBLICATION						
7	THENJOT DOWN THE DOPE ABOUT IT ON THIS FORM.												
	CAREFULLY TEAR IT OUT, FOLD IT AND DROP IT IN THE MAIL. DATE SENT												
LID DROS II EV IIIE MAILE.													
PUBLICAT	FION NUMBE	₽			PUBLICATION D	ATE	PUBLICATION TITLE						
BE EXACT PIN-POINT WHERE IT IS IN THIS SPA						LL WHA	AT IS WRONG						
PAGE NO.	PAGE PARA- FIGURE TABLE A				AND WHAT SHOULD BE DONE ABOUT IT.								
PRINTED	NAME, GRA	DE OR TITL	E AND TELE	PHONE NU	MBER	SIGN HE	RE						

DA 1 JUL 79 2028-2

PREVIOUS EDITIONS ARE OBSOLETE. P.S.--IF YOUR OUTFIT WANTS TO KNOW ABOUT YOUR RECOMMENDATION MAKE A CARBON COPY OF THIS AND GIVE IT TO YOUR HEADQUARTERS.

THE METRIC SYSTEM AND

EQUIVALENTS

Linear Measure

- 1 centimeter = 10 millimeters = .39 inch
- 1 decimeter = 10 centimeters = 3.94 inches
- 1 meter = 10 decimeters = 39.37 inches
- 1 dekameter = 10 meters = 32.8 feet
- 1 hectometer = 10 dekameters 328.08 feet
- 1 kilometer = 10 hectometer 3280.8 feet

Weights

- 1 centigram = 10 milligrams = .15 grain
- 1 decigram = 10 centigrams = 1.54 grains
- 1 gram = 10 decigrams = .035 ounce
- 1 dekagram = 10 grams = .35 ounce
- 1 hectogram = 10 dekagrams = 3.52 ounces
- 1 kilogram = 10 hectograms = 2.2 pounds
- 1 quintal = 100 kilograms = 220.46 pounds
- 1 metric ton = 10 quintals = 1.1 short tons

Liquid Measure

- 1 centiliter = 10 milliliters = .34 fl. ounces
- 1 deciliter = 10 centiliters = 3.38 fl. ounces
- 1 liter = 10 deciliters = 33,81 fl. ounces
- 1 dekaliter = 10 liters = 2.64 gallons
- 1 hectoliter = 10 dekaliters = 26.42 gallons
- 1 kiloliter = 10 hectoliters = 264,18 gallons

PIN: 058973-000

This fine document...

Was brought to you by me:



<u>Liberated Manuals -- free army and government manuals</u>

Why do I do it? I am tired of sleazy CD-ROM sellers, who take publicly available information, slap "watermarks" and other junk on it, and sell it. Those masters of search engine manipulation make sure that their sites that sell free information, come up first in search engines. They did not create it... They did not even scan it... Why should they get your money? Why are not letting you give those free manuals to your friends?

I am setting this document FREE. This document was made by the US Government and is NOT protected by Copyright. Feel free to share, republish, sell and so on.

I am not asking you for donations, fees or handouts. If you can, please provide a link to liberatedmanuals.com, so that free manuals come up first in search engines:

Free Military and Government Manuals

- SincerelyIgor Chudovhttp://igor.chudov.com/
- Chicago Machinery Movers